## Non-normality in combustion-acoustic interaction in diffusion flames: a critical revision

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Perturbations in a non-normal system can grow transiently even if the system is linearly stable. If this transient growth is sufficiently large, it can trigger self-sustained oscillations from small initial disturbances. This has important practical consequences for combustion–acoustic oscillations, which are a persistent problem in rocket and aircraft engines. Balasubramanian & Sujith (*J. Fluid Mech.*, vol. 594, 2008, pp. 29–57) modelled an infinite-rate chemistry diffusion flame in an acoustic duct and found that the transient growth in this system can amplify the initial energy by a factor,  $G_{max}$ , of the order of  $10^5$  to  $10^7$ . However, recent investigations by L. Magri and M. P. Juniper have brought to light certain errors in that paper. When the errors are corrected,  $G_{max}$  is found to be of the order of 1 to 10, revealing that non-normality is not as influential as it was thought to be.

Key words: Acoustics, Combustion, Flames

## 1. Results and discussion

Recent investigations have brought to light certain errors in Balasubramanian & Sujith (2008, labelled B&S in this note). We use the same model, discretization and non-dimensionalization as in B&S. The required corrections to B&S are listed below.

(a) The analytical steady solution,  $Z_{st}$  (appendix B, p. 54), obtained by separation of variables, is

$$Z_{st} = X_i(1-\alpha) - Y_i\alpha - \frac{2}{\pi}(X_i + Y_i) \sum_{n=1}^{+\infty} \frac{\sin(n\pi\alpha)}{n(1+b_n)} \cos(n\pi y_c) (e^{a_{n1}x_c} + b_n e^{a_{n2}x_c}), \quad (1.1)$$

where

$$a_{n1} \equiv \frac{Pe}{2} - \sqrt{\frac{Pe^2}{4} + n^2\pi^2}, \quad a_{n2} \equiv \frac{Pe}{2} + \sqrt{\frac{Pe^2}{4} + n^2\pi^2},$$
 (1.2)

$$b_n \equiv -\frac{a_{n1}}{a_{n2}} \exp\left(-2L_c \sqrt{\frac{Pe^2}{4} + n^2 \pi^2}\right).$$
 (1.3)

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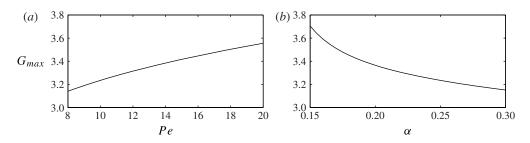


FIGURE 1. The growth factor,  $G_{max}$ , as a function of (a) the Péclet number, Pe, and (b) the fuel slot half-width,  $\alpha$ . Panel (a) has  $\alpha = 0.25$ , and (b) has Pe = 10.

The non-dimensional coordinates of the combustion domain are  $x_c$  and  $y_c$ .

- (b) The expressions for  $C_m^{(n)}$  and  $W_{mk}$  (equation (7), p. 36) are  $2/L_c$  times the original terms, due to the Galerkin projection. Furthermore,  $W_{mk} = 1/L_c$  when k = m.
- (c) The variable  $Y_1$  on the right-hand side of the terms  $R_{nm}$  and  $J_{nm}$  (equation (13), p. 37) is  $Y_i$ .
- (d) The variable  $\dot{Q}_{av}$  (equations (18) and (19), p. 39) is to be divided by 2 due to non-dimensionalization over the cross-sectional area.
- (e) The multiplying factor in front of matrix  $\mathbf{M}_1$  (appendix B, p. 54) is  $1/((T_i + T_{ad})/2)$ .
- (f) The expression for matrix  $\boldsymbol{B}_{NN}$  (appendix B, p. 54) is  $\boldsymbol{B}_{NN} = -\boldsymbol{D} + \boldsymbol{A}_1 \boldsymbol{A}_2 + \boldsymbol{A}_3 \boldsymbol{A}_4 + \boldsymbol{A}_5$ , where

$$\mathbf{A}_{5} = \frac{1}{(T_{i} + T_{ad})/2} \begin{bmatrix} 0 \ 0 \ 0 \ \dots \ 0 \ \sin(\pi x_{f}) \ \sin(2\pi x_{f}) \ \dots \ \sin(K\pi x_{f}) \end{bmatrix}^{\mathrm{T}} \times \begin{bmatrix} J_{00} \ \dots \ J_{0M} \ 0 \ \dots \ 0 \end{bmatrix}$$
(1.4)

and K is the number of Galerkin modes for acoustic discretization.

- (g) The damping terms in the matrix **S** (appendix B, p. 55) are  $+2\pi\xi_1$ ,  $+4\pi\xi_2$ , ...,  $+2K\pi\xi_K$ .
- (*h*) The numerator of matrix  $A_4$  (appendix B, p. 55) is 1 due to non-dimensionalization over the cross-sectional area of the duct.

We perform computations with  $50 \times 50$  Galerkin modes in the flame domain and six modes in the acoustic domain. When we increase the number of Galerkin modes to  $70 \times 70$  in the flame and 12 in the acoustics, the eigenvalues and singular values change by less than 15%. The fixed parameters are: the fuel mass ratio,  $Y_i = 3.2$ ; the oxidizer mass ratio,  $X_i = 3.2/7$ ; and the average temperature,  $T_{av} = 1/0.685$ . We set the damping coefficients to  $c_1 = 0.013$  and  $c_2 = 0.08$  in order to have marginally stable systems. The nonlinear behaviour of this thermo-acoustic system is not considered because it has been fully characterized by Illingworth, Waugh & Juniper (2013).

Figure 1 shows the growth factor,  $G_{max}$ , as a function of (*a*) the Péclet number, *Pe*, and (*b*) the non-dimensional half-width of the fuel slot,  $\alpha$ . (Note that we use the same norm as B&S, even though Chu's norm would be a more appropriate measure of the energy (Chu 1965).) In both cases,  $1 < G_{max} \leq 10$ . These plots can be compared with figures 9 and 10 in B&S. Figure 2 shows the eigenvalues and the pseudospectra for this thermo-acoustic system. These can be compared with figure 11 in B&S. The pseudospectra around the most unstable eigenvalues are nearly concentric circles whose values decrease rapidly as the distance from the eigenvalue increases. This is a further demonstration that the system is only weakly non-normal,

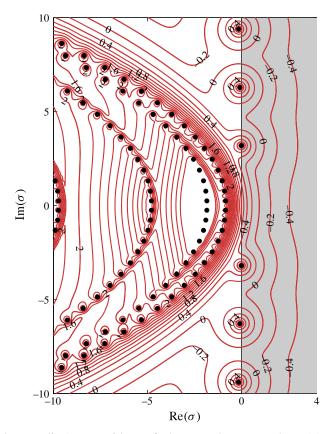


FIGURE 2. (Colour online) Logarithm of the pseudospectra,  $\log_{10}(\epsilon)$ . The parameters are the same as figure 1, with  $\alpha = 0.25$  and Pe = 10. The dominant eigenvalue is  $\sigma = -0.003 \pm 3.193i$ .

because a marginally stable but highly non-normal system would have pseudospectra that protrude significantly into the unstable half-plane (Trefethen & Embree 2005). Nevertheless, it is worth noting that Juniper (2011) showed that even a small amount of non-normality can make a system somewhat more susceptible to triggering.

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